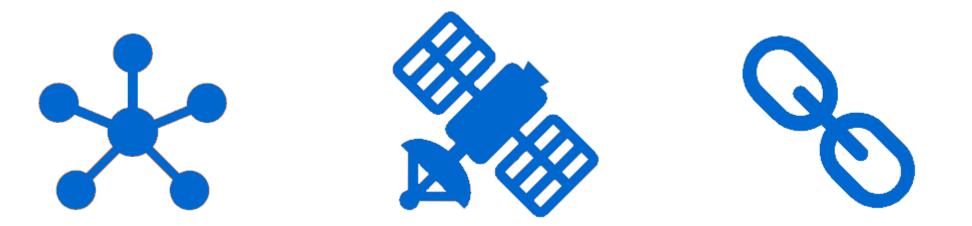
Proposal for a Geostationary Microwave Amateur Radio Payload

Kai Siebels, DH0SK Matthias Bopp, DD1US





Considerations



Orbit Satellite & Platform Payload













Orbit – GEO pro's and con's

Among the possible orbits such as LEO, MEO, HEO and GEO, the **GEO** orbit has proven to be the <u>most suitable</u> orbit for the requirements based on the extensive experience with OSCAR-100.

Advantages:

- + Great coverage (a GEO satellite can see the surface of the Earth up to 81 degrees away from its sub-satellite-point).
- + No problems on the ground with tracking using high-gain antennas with amateur equipment
- + Reliable 24/7 availability even in the case of emergency radio applications for radio amateurs
- + Relaxed requirements with respect to radiation hardness as outside of Van Allan belt

Disadvantages:

- Orbit position (ITU compliant) depending on the host (co-flight)
- Hosted payload (limited control possibilities and telemetry/science data)
- Not all areas of the world can be reached. If possible an orbit position should be chosen which covers a high population of radio amateurs

A HEO orbit would be second best and still a viable option.

Orbit – positions to reach many radio amateurs



Elevation contours 10deg increments (Source: AMSAT-UK)

Coverage from a GEO satellite at 5 degrees west. Not ideal because the area of North America / Canada is not properly covered



Coverage from a GEO satellite located at 47 degrees west. Almost ideal, with the disadvantage that some eastern European countries and Asia is not covered. North America and most of Canada with a large number of active radio amateurs are covered.

<u>Good compromise would be a position at ~43 degrees west</u> which supports also the Eastern European countries and still most of North America including Canada





Satellite and Platform

With regard to the costs for satellite platforms and also launching a satellite, the following must be taken into account for an amateur radio satellite:

Hosted Payload (fly with a provider who also determines the orbit position):

	Small GEO chemical	Small GEO hybrid	Small GEO electrical	Micro GEO electrical
estimated Payload Power	~2,5kW	~3,6kW	~9,5kW	~1,5kW
estimated Gbps/BW	~25	~35	~100	~5
Payload Mass		~400kg	~500kg	~150kg
AMSAT-DL				

Satellite and Platform

Advantages of a solution based on a hosted payload:

- + With OSCAR-100, radio amateurs can draw on experience in operating a payload as a hosted payload.
- + From a cost perspective, this is an ideal way to fly a small payload.
- + There is a relatively large amount of DC power available
- + The radio amateurs have no effort when it comes to, for example, attitude control and management & control (Launch-And-Early-Orbit-Phase LEOP, In-Orbit-Testing IOT) of the satellite.

Disadvantages of a solution based on a hosted payload:

- Orbit position cannot be freely chosen (given by provider)
- No influence on steering and maneuvers. (Transponders are simply turned on or off)
- Any changes to the settings (LCAMPS gain, FGM. ALC, etc.) must be adjusted via the Satellite-Communications-Controller (SCC).
- Difficulties with AMSAT's own developments (qualification, influencing the main payload).
- No own telecommand/telemetry for control and monitoring purposes (possibly solvable with own OBP).
- No real time access and control of experiments (e.g. from schools / universities).



Satellite and Platform

Own AMSAT (amateur radio) mission based on a Micro GEO supported by ESA:

- An ideal solution would be a Micro GEO satellite for amateur radio. As described above, we are in the ~1.5kW/150kg class, which would be ideal for a payload for amateur radio.
- Small electric powered engines make it possible to adjust the satellite's orbit position if necessary. All
 areas of the world can be reached here. The small satellite can also be used easily in the event of an
 emergency radio situation.
- A development of such a **Micro GEO Platform** in an e.g. ESA program (e.g. by industry with the support of AMSAT for a suitable payload) would be an ideal situation.
- This approach also supports AMSAT's goals for training, science and development as well as the opportunity for industry to develop and test a Micro GEO Platform and to inspire young people for technology through the free frequencies of amateur radio.

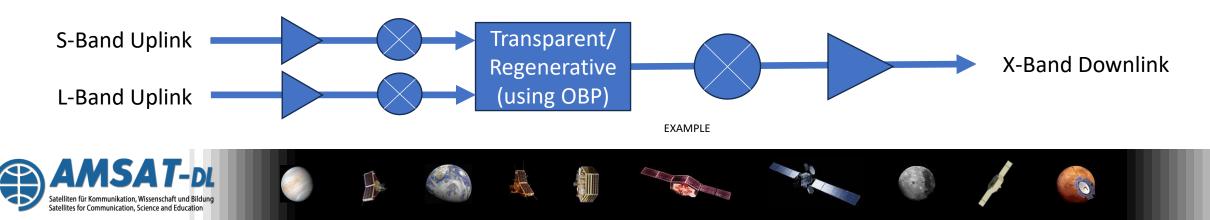


Payload ideas

A payload for amateur radio should allow the greatest possible scope for experiments in different bands. In addition, the payload should also be cost/benefit optimized and thus keep the effort on the satellite as low as possible. It also makes sense to use modules that are compatible with the SGEO platform (100V bus, CAN bus, HPC, HVC).

A main downlink in the **10GHz band (3cm)** is essentially suitable for narrow and broadband applications. Reception in the 10GHz band makes it relatively easy as there is a wide range of converters, LNBs and antennas on the market. You can also access a broad wealth of experience from OSCAR-100 and use existing receiving stations.

A wide range of amateur frequency bands could be used for the uplink, which allows a wide spectrum for experiments on the ground. Crossband operation such as **23cm uplink** and **13cm uplink** with a common downlink on the **3cm band** would be conceivable.



Payload ideas – feasible frequency bands

The following frequency bands are available for amateur radio via satellites:

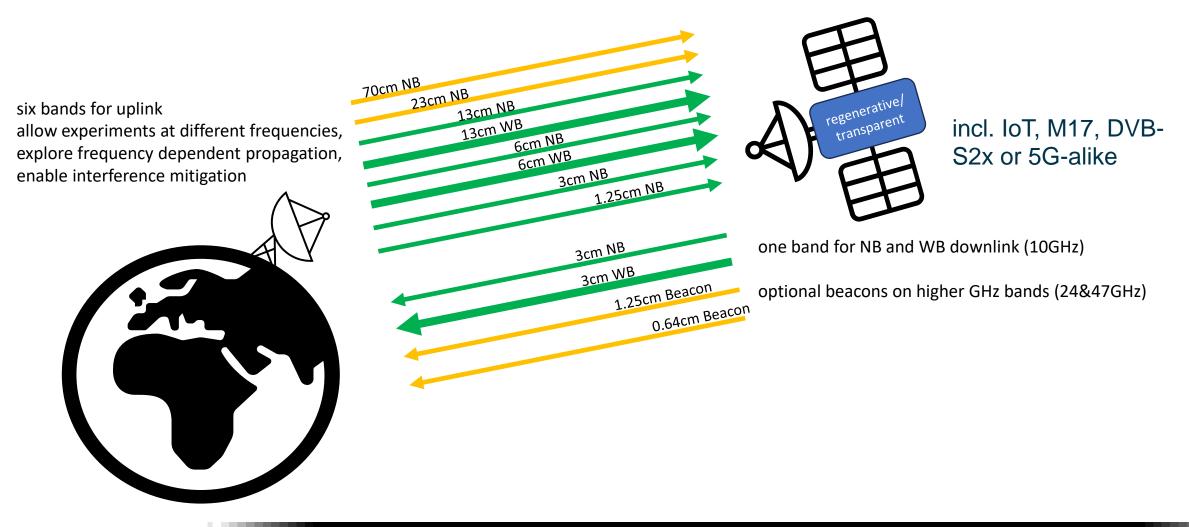
Downlink NB	Uplink NB	Accommodation possible	Remarks
3cm Band (higher section)	144–146 MHz / 2m Band	red	antenna size with reasonable gain very large
3cm Band (higher section)	435–438 MHz / 70cm Band	orange	antenna size with reasonable gain large, t.b.c.
3cm Band (higher section)	1 260–1 270 MHz / 23cm Band	orange	if band still available most countries
3cm Band (higher section)	2 400–2 450 MHz / 13cm Band	green	like OSCAR-100
3cm Band (higher section)	3402–3410 MHz / 9cm Band	red	only Space-to-Earth and not in all regions available
3cm Band (higher section)	5 650–5 670 MHz / 6cm Band	green	accommodation of antennas feasible
10,450-10,500 GHz / 3cm Band (higher section)	10,450-10,500 GHz / 3cm Band (lower section)	green	feasible with a suitable duplexer
3cm Band (higher section)	24–24,05 GHz Band / 1,25cm Band	green	horn antenna / accommodation ok /maybe beacon
3cm Band (higher section)	47–47,2 GHz Band	orange	horn antenna / difficult for users, maybe beacon
3cm Band (higher section)	76–77,5 GHz Band	red	horn antenna / difficult for users

Advantage: no ITU regulation needed

Remark: to cover the whole visible footprint of the satellite an antenna gain of approx. 17dBi is needed

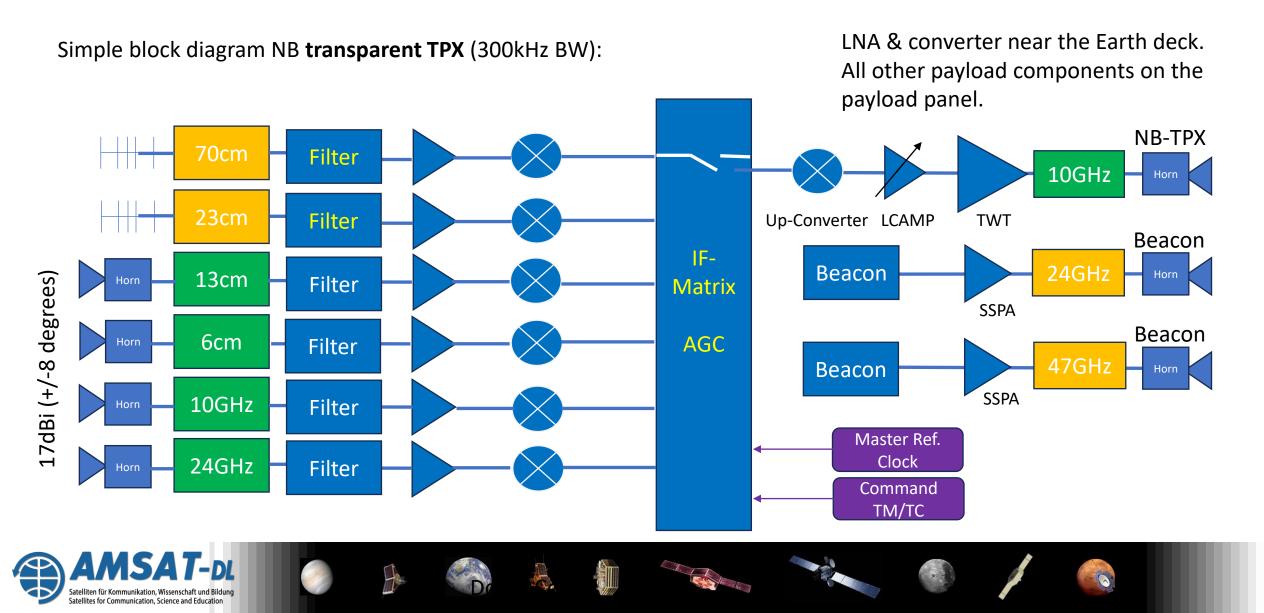


Payload ideas – feasible frequency bands





Payload ideas – transparent transponder

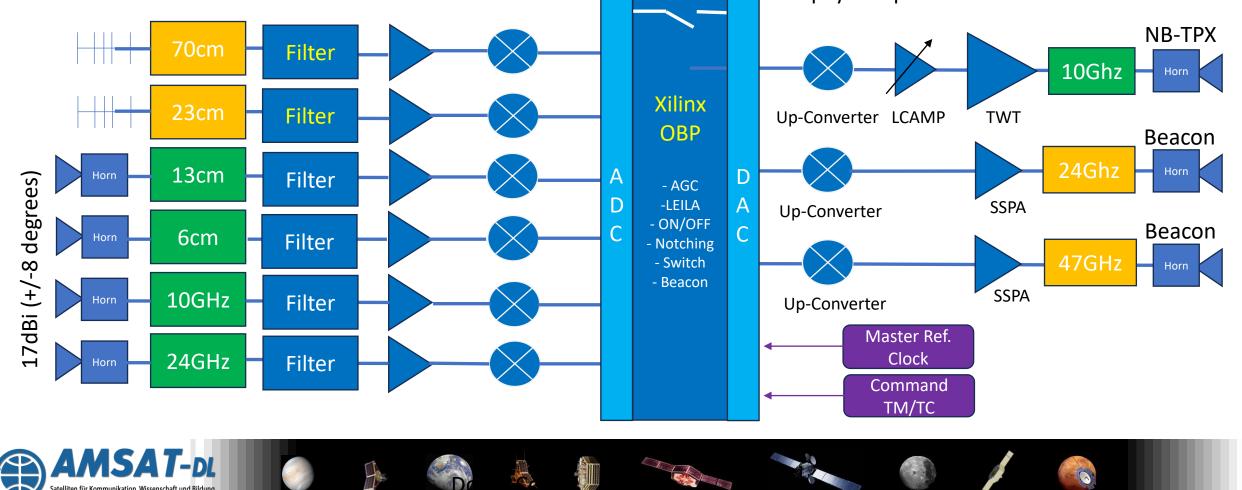


Payload ideas – regenerative transponder

Simple block diagram NB regenerative TPX (300kHz BW):

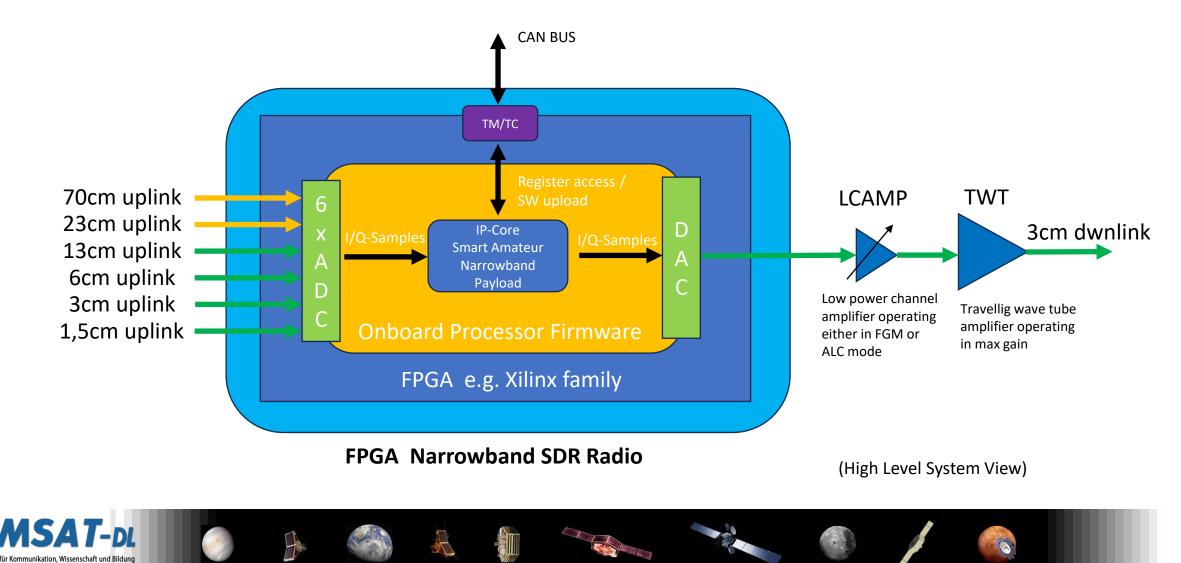
Satellites for Communication, Science and Education

LNA & converter near the Earth deck. All other payload components on the payload panel.

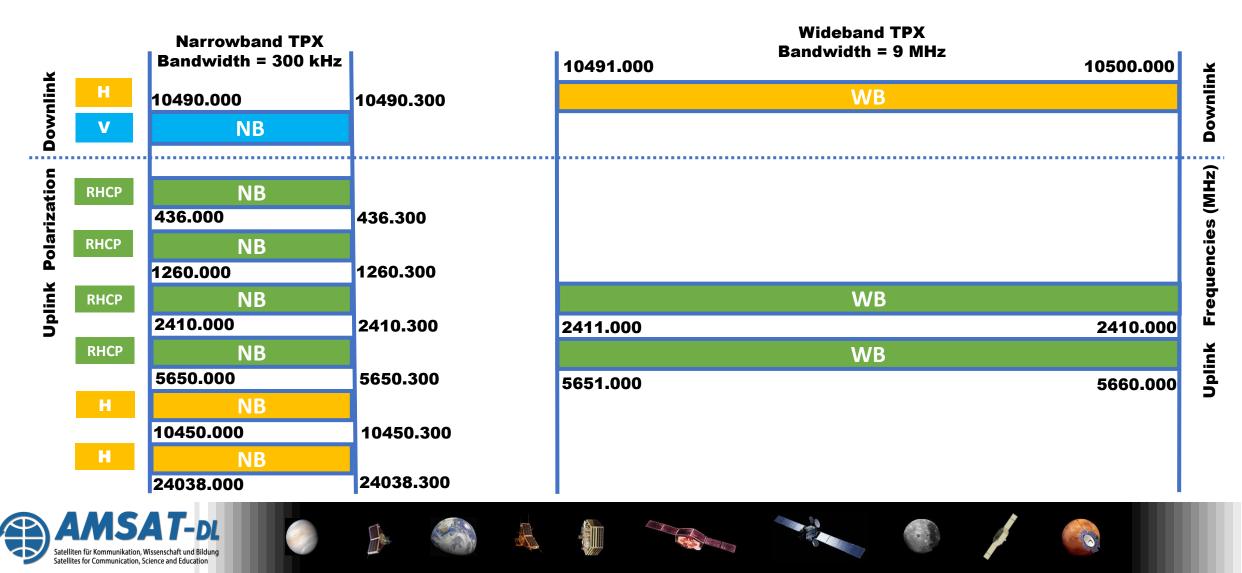


Payload ideas – block diagram processor

Satellites for Communication, Science and Education



Payload ideas - draft frequency plan



Payload ideas - link budget 13cm/3cm TPX

- SX-NB-transponder
- Uplink: 2.4GHz / 13cm band
- Downlink: 10GHz / 3cm band
- GEO satellite position: -43° West
- Ground station position: San Francisco (at the very edge of the footprint)

ι	Jplink
TX ground terminal	
Freq	2.4 GHz
Wavelength	0.125 m
Dish size	0.8 m
Aperture efficiency	0.65
Gain	24.2 dBic
Beamwidth	10.9 deg
PA output	5 W
TX losses	3.0 dB
EIRP	28.2 dBW
Path	
TX lat	37.7562 deg N
TX lon	-122.443 deg E
Slant range	41720 km
Path loss	192.5 dB
Atmospheric losses	0 dB
Fade margin	1 dB
Transponder input Spacecraft G/T	-12 dB/K
C/No	51.3 dBHz
Channel BW	2400 Hz
C/N	17.5 dB
<u>Pointing data TX</u> Azimuth Elevation	96.5 deg 0.4 deg
Lievation	0.4 ueg

Do	wnlink
Transponder output	
Freq	10.5 GHz
Wavelength	0.029 m
TWTA output	100 W
Output back off	6 dB
TX losses	1.5 dB
Ant gain	17 dBi
EIRP	29.5 dBi
Power sharing	50 channe
EIRP per channel	12.5 dBW
Path	
RX lat	37.7562 deg N
RX lon	-122.443 deg E
Slant range	41720 km
Path loss	205.3 dB
Atmospheric losses	1 dB
Fade margin	1 dB
<u>RX ground terminal</u> Dish size	0.8 m
Aperture efficiency	0.65
Gain	37.0 dBi
Beamwidth	2.5 deg
Antenna noise temp	45 K
LNA noise figure	1.2 dB
LNA noise temp	92.3 K
Ground terminal G/T	15.6 dB/K
C/No	49.5 dBHz
Channel BW	2400 Hz
C/N per user	15.7 dB
Pointing data RX	
Azimuth	96.5 deg
Elevation	0.4 deg

Source Spreadsheet: GOTLE















Payload ideas - link budget 6cm/3cm TPX

- SX-NB-transponder
- Uplink: 5.7GHz / 6cm band
- Downlink: 10GHz / 3cm band
- GEO satellite position: -43° West
- Ground station position: San Francisco (at the very edge of the footprint)

Upl	link
TX ground terminal	
Freq	5.6 GHz
Wavelength	0.053571429 m
Dish size	0.8 m
Aperture efficiency	0.65
Gain	31.6 dBic
Beamwidth	4.7 deg
PA output	10 W
TX losses	3.0 dB
EIRP	38.6 dBW
Path	
TX lat	37.7562 deg N
TX lon	-122.443 deg E
Slant range	41720 km
Path loss	199.8 dB
Atmospheric losses	1 dB
Fade margin	1 dB
Transponder input	
Spacecraft G/T	-12 dB/K
C/No	53.3 dBHz
Channel BW	2400 Hz
C/N	19.5 dB
Pointing data TX	
	96.5 deg
Azimuth	90.5 ueg

Do	wnlink		
Transponder output		1	
Freq	10.5	GHz	
Wavelength	0.029		
TWTA output	100	W	
Output back off		dB	
TX losses	1.5		
Ant gain		dBi	
EIRP	29.5	1	
Power sharing		channels	
EIRP per channel	12.5	dBW	
Death			
Path	27 75 62		
RX lat	37.7562	-	
RX lon	-122.443 41720	-	
Slant range Path loss	205.3		
Atmospheric losses		dB	
Fade margin	1	dB	
RX ground terminal			
Dish size	0.8	m	
Aperture efficiency	0.65		
Gain	37.0	dBi	
Beamwidth		deg	
Antenna noise temp	45	, Ŭ	
LNA noise figure	1.2	dB	
LNA noise temp	92.3	к	
Ground terminal G/T	15.6	dB/K	
C/No	49.5	dBHz	
Channel BW	2400	Hz	
C/N per user	15.7	dB	
Pointing data RX			
Azimuth	96.5	U	
Elevation	0.4	deg	

Source Spreadsheet: GOTLE

Payload ideas - link budget 1.25cm/3cm TPX

- KX-NB-transponder
- Uplink: 24GHz / 1.25cm band
- Downlink: 10GHz / 3cm band
- GEO satellite position: -43° West
- Ground station position: San Francisco (at the very edge of the footprint)

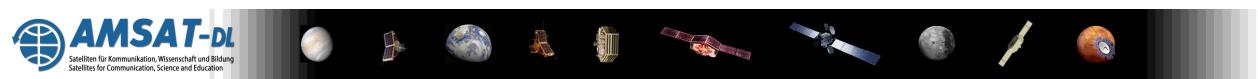
U	plink
TX ground terminal	
Freq	24 GHz
Wavelength	0.0125 m
Dish size	1.2 m
Aperture efficiency	0.65
Gain	47.7 dBic
Beamwidth	0.7 deg
PA output	2 W
TX losses	3.0 dB
EIRP	47.7 dBW
<u>Path</u>	
TX lat	37.7562 deg N
TX lon	-122.443 deg E
Slant range	41720 km
Path loss	212.5 dB
Atmospheric losses	1 dB
Fade margin	1dB
<u>Transponder input</u> Spacecraft G/T	-12 dB/K
C/No	49.9 dBHz
Channel BW	2400 Hz
C/N	16.1 dB
Pointing data TX	
Azimuth	96.5 deg
Elevation	0.4 deg

Do	ownlink
Transponder output	
Freq	10.5 GHz
Wavelength	0.029 m
TWTA output	100 W
Output back off	6 dB
TX losses	1.5 dB
Ant gain	17 dBi
EIRP	29.5 dBi
Power sharing	50 channels
EIRP per channel	12.5 dBW
Path	
RX lat	37.7562 deg N
RX lon	-122.443 deg E
Slant range	41720 km
Path loss	205.3 dB
Atmospheric losses	1 dB
Fade margin	1dB
RX ground terminal	
Dish size	0.8 m
Aperture efficiency	0.65
Gain	37.0 dBi
Beamwidth	2.5 deg
Antenna noise temp	45 K
LNA noise figure	1.2 dB
LNA noise temp	92.3 K
Ground terminal G/T	15.6 dB/K
C/No	49.5 dBHz
Channel BW	2400 Hz
C/N per user	15.7 dB
Pointing data RX	
Azimuth	96.5 deg
Elevation	0.4 deg
	0.1.005

Source Spreadsheet: GOTLE

Payload ideas - summary

- All suggested band / NB transponder combinations are feasible with very reasonable effort at the ground station (comparable to QO-100).
- For the WB transponder (S-band uplink and X-band downlink) the effort will be identical to QO-100 (with the option for a second WB uplink in the 6cm band).



Payload ideas – other experiments

An own AMSAT (amateur radio) mission based on a Micro GEO supported by ESA provides opportunities for multiple additional experiments supporting AMSAT's goals for training, science and development to inspire young people for technology through the free frequencies of amateur radio.

This could be an excellent platform for further cooperation with organisations like ESERO: "Interactive" experiments of students would feasible as AMSAT could control them in real time and provide the results to the students such as data from cameras and sensors.

Such experiments are normally difficult to be qualified for commercial satellites (when using them as a hosted payload) but in the case of an own GEO such requirements could be relaxed.

Finally such a mission could provide an excellent platform for disaster / emergency communications directly via the GEO satellite transponders (as demonstrated in Turkey during the Earthquake 2023). Using the Micro GEO electrical propulsion system the GEO satellite could be moved to cover respective areas impacted by disasters on short notice.





Thank you very much !













